

# Near Earth Asteroids Prospection, Orbit Modification and Mining

by

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### Asteroid Mining : challenge of the future

• Geochemical Groups for meteorites and asteroids with typical elements occuring in mineral associations:

Group Elem		Elements (selection)
•	Siderophile	<u>Fe, Ni,</u> Co, Cu, <u>Au</u> , Pd, <u>Pt,</u> Os, Ir
•	Chalcophile	<u>Fe, Ag</u> , Cd, In, Th, Pb, Bi, S, Se, Te
•	Lithophile	Rb, Cs, Be, <u>Al</u> , Sc, Th, U, <u>Ti</u> , Nb, Ta, Cr, Mn, <u>rare -Earth elements</u>

- How to use these resources for the benefit of mankind ?
- First step: sending robotic probes to Near Earth Asteroids, preferably <u>Potentially Hazardous Asteroids</u>



### Potentially Hazardous Asteroids (PHAs)

•Minimum intersection distance less than 0.05 AU (approx. 7 500 000 km)

•Velocity change ( $\Delta$  V) for a spaceship is below 12 km/s to reach the PHA

•Size distribution of PHAs depending on their diameter D (rough estimation) :



### What to do with really Hazardous Asteroids ?

#### <u>Deflect</u> the asteroid by various techniques:

- kinetic impact
- lateral (nuclear) detonation
- connecting rocket engines (a Space Tug) to the Asteroid

#### Modify the orbit of the asteroid:

- -from current Solar orbit to a stable Earth orbit beyond the Moon
- connecting a Space Tug with advanced propulsion systems to "catch and "guide" the asteroid



### The Arrival of the Asteroid

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take 2008EV5

#### Potential Candidates of PHAs for Orbit Modification and Mining

List of PHAs with their physical and orbital parameters. For the calculated mass value a spherical shape with homogeneous bulk density of 3 g/cm<sup>3</sup> has been assumed.

Asteroid	Diameter (m)	Mass (kg)	Semi-major axis (AU)	Eccentricity	Spectral Type	
2004 MN4	270	3.092x10 <sup>10</sup>	0.922	0.191	Sq	
1982 DB	330	5.645x10 <sup>10</sup>	1.489	0.360	Xe	
1998SF36	330	5.645x10 <sup>10</sup>	1.324	0.280	S	We take 2008E as an example
2005YU55	400	1.005x10 <sup>11</sup>	1.157	0.430	С	
2008 EV5	450	1.431x10 <sup>11</sup>	0.958	0.084	S	
1982 XB	500	1.963x10 <sup>11</sup>	1.835	0.446	S	
1999RQ36	493	1.882x10 <sup>11</sup>	1.126	0.204	С	



What is the estimated <u>Transfer Energy</u> (TE) to change the current Solar orbit of the asteroid 2008 EV5 into an Earth orbit ?

- We take the <u>difference in Kepler Energy</u> (ΔK) between the asteroid's current Solar orbit (K<sub>1</sub>, a<sub>1</sub>) and Earth's orbit (K<sub>2</sub>, a<sub>2</sub>) :

 $\Delta K = K_2 - K_1 = GM/2 \cdot (1/a_1 - 1/a_2)$  (Roy 1988)

G Gravitation constant M Mass of the Sun a semi-axis

 $\Delta K = (6.67 \times 10^{-11} \cdot 1.99 \times 10^{30})/2 \cdot (1/1.433 \times 10^{11} - 1/1.496 \times 10^{11}) = \frac{1.941 \times 10^{7}}{J/kg}$ 

- We multiply it with the mass (m) of 2008 EV5 : 1.431x10<sup>11</sup> kg

 $\Delta K \cdot m = TE$  (Joule) 1.941 x10<sup>7</sup> · 1.431 x10<sup>11</sup> = 2.778 x10<sup>18</sup> J

The required Transfer Energy is approx. <u>2.778 Exajoule</u> ( 2.778 million Terajoule) We need an advanced propulsion system :

e.g. the <u>BUSSARD FUSION ENGINE</u>, also called the "quietelectric- discharge (QED) engine" (Robert W. Bussard, 1997, 2002)

- Deuterium and Helium-3 are fusing to Helium-4 plus protons

 $_{2}H + _{3}He = _{4}He + p$ 

- Each reaction releases an energy of approx. 18.3 MeV

- For the Deuterium-Helium-3 fuel R. Bussard gives a specific energy of 3.5 x10<sup>14</sup> J/kg

-For the 2008 EV5 maneuver we can compute the required fuel mass:

 $2.778 \times 10^{18} (J) / 3.5 \times 10^{14} (J/kg) = 7937 kg$  (8 tons of fuel) (Transfer Energy)

- Helium-3 can be found on the Lunar surface as a product of the Solar wind.



#### Asteroid Space Tugs (unmanned)

The tugs and the Bussard Fusion Engines have to be produced in series production.

The goal is to "catch" a number of asteroids within some decades.



#### Asteroid 2008 EV5, guided by Space Tugs

The tugs can apply forces in every direction
Tug 1 adjusts the fligt track by short engine thrusts
Tug 2 applies the primary force for the orbital maneuvers



#### Mining the Asteroid :

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Once stabilized in an Earth orbit beyond the Moon, the mining process is started along the major axis of the asteroid.

A Manned Mining Station is docked to the asteroid.





#### Advantages of an asteroid's Earth orbit :

- Short flight time from Earth (5 to 7 days)
- Rate of <u>mining advance</u>, removal of ore and storage can be kept equal to the rate of <u>cargo shipping</u>
- After mining : the hollow asteroid can be used as a <u>shelter</u> for <u>industrial facilities</u> or <u>storage</u> (water, oxygen, etc.) for industry in space
- Asteroids with more than 400 m in diameter can be used for <u>Rotating Human Colonies</u> with artificial gravity.
- The remaining stony crust provides shelter against cosmic rays, Solar flares and meteorites

#### Asteroid Colony (sections)

The rotating torus inside provides artificial gravity. Natural sunlight is collected by parabolic mirrors and beamed into the cave.





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How to apply an energy of 2.778 x10<sup>18</sup> Joule to an asteroid of 450 m in diameter ?

- connect a spaceship (space tug) with usual LOX/LOH rocket engines?
- Specific Energy (*calorific value*) of Hydrogen :  $1.418 \times 10^8$  J/kg 2.778 x10<sup>18</sup> (J) / 1.418 x10<sup>8</sup> (J/kg) =  $1.96 \times 10^{10}$  kg of H<sub>2</sub>

-We would need a fuel mass of 20 million tons of Hydrogen !

- Is this feasable ??



#### Supplementary sheet 2:

### The Mining Process :

- The asteroid is excavated up to 50 % of its volume
- Heavy elements are supposed to be in its core
- The excavated cave is filled with a pressurized gas
- The muck (the rock chips) are removed in a vacuum conveyor tube to the Mining Station
- The raw material is processed and stored in the Manned Mining Station
- Cargo ships transport the material to Low Earth Orbit or to the Lagrange Points for further industrial use, e.g. in metallurgical plants



## Thank You for Your attention !

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Drawings by C. Böck